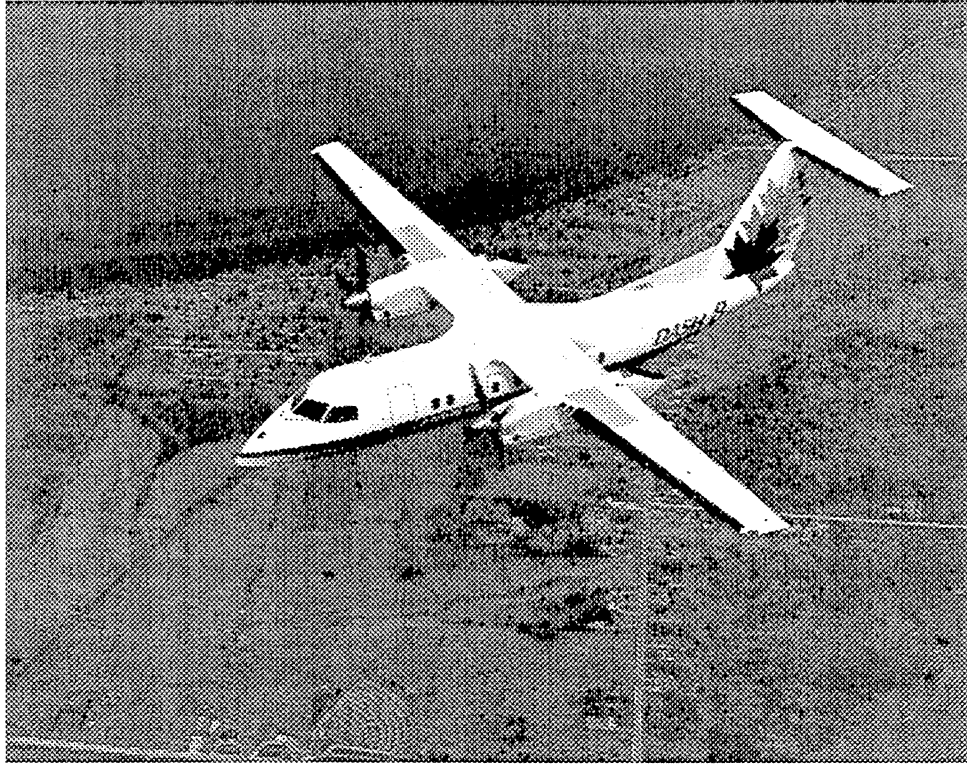


Boeing Canada  
de Havilland Division

PSM 1-8-14

**DASH 8 CRASH-FIRE-RESCUE INFORMATION**



**DASH 8  
SERIES 100  
CRASH-FIRE-RESCUE  
INFORMATION**

**PSM 1-8-14**

**THIS MANUAL IS FOR THE INFORMATION  
OF FIRE AND RESCUE PERSONNEL.**

**PRODUCT SAFETY DEPARTMENT OF CUSTOMER SUPPORT  
BOEING OF CANADA LTD.  
de HAVILLAND DIVISION  
DOWNSVIEW, ONTARIO, CANADA M3K 1Y5**

**APRIL 1985  
RE-ISSUED SEPTEMBER 30, 1991**



**Boeing Canada**  
de Havilland Division**DASH 8 CRASH-FIRE-RESCUE INFORMATION****LIST OF EFFECTIVE PAGES**

Insert this page and latest revised page(s). Destroy superseded pages. Listed below are all current pages of the manual and their dates of issue.

<b>PAGE</b>	<b>DATE</b>
Title Page	September 30, 1991
A	September 30, 1991
Log of Revisions	September 30, 1991
Table of Contents	September 30, 1991
1	September 30, 1991
2	September 30, 1991
3	September 30, 1991
4	September 30, 1991
5	September 30, 1991
6	September 30, 1991
7	September 30, 1991
8	September 30, 1991
9	September 30, 1991
10	September 30, 1991
11	September 30, 1991
12	September 30, 1991
13	September 30, 1991
14	September 30, 1991
15	September 30, 1991
16	September 30, 1991
17	September 30, 1991
18	September 30, 1991
19	September 30, 1991
20	September 30, 1991
21	September 30, 1991
22	September 30, 1991





**Boeing Canada**  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

**THIS PAGE INTENTIONALLY LEFT BLANK**

**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

**TABLE OF CONTENTS**

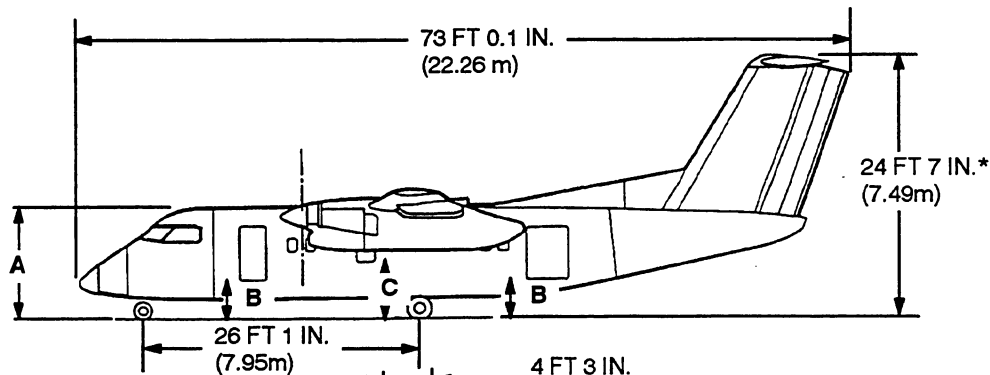
	PAGE
GENERAL ARRANGEMENT .....	1
FAMILIARIZATION AND LOCATION GUIDE .....	2
EXTERIOR WALK-AROUND .....	3
INTERIOR ARRANGEMENT .....	4
CABIN CROSS-SECTION .....	5
FLIGHT COMPARTMENT (VIEW FORWARD) .....	6
FLIGHT COMPARTMENT (VIEW AFT) .....	7
EXITS AND SERVICE DOORS .....	8
AIRCRAFT DOORS AND GROUND SERVICE PANELS .....	9
EVACUATION ROUTES .....	10
PASSENGER AND CREW ESCAPE SYSTEMS .....	11
BAGGAGE DOOR OPERATION .....	12
FLIGHT COMPARTMENT EMERGENCY HATCH .....	13, 14
ENGINE DANGER AREAS .....	15
CUT-THROUGH AREAS .....	16
FUSELAGE SAFETY EQUIPMENT LOCATIONS .....	17
CREW OXYGEN LOCATIONS .....	18
FIRE CONTROL RECOMMENDATIONS .....	19
FLAMMABLE MATERIAL LOCATIONS .....	20
ENGINE FIRE ACCESS LOCATIONS .....	21
ENGINE FIRE EXTINGUISHER AND BATTERY POWER SWITCH LOCATION .....	22

**Boeing Canada**  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

**THIS PAGE INTENTIONALLY LEFT BLANK**



**DASH 8 CRASH-FIRE-RESCUE INFORMATION**



**WHEELS EXTENDED**

**WHEELS RETRACTED**

A=125.00 IN.(3.18m)

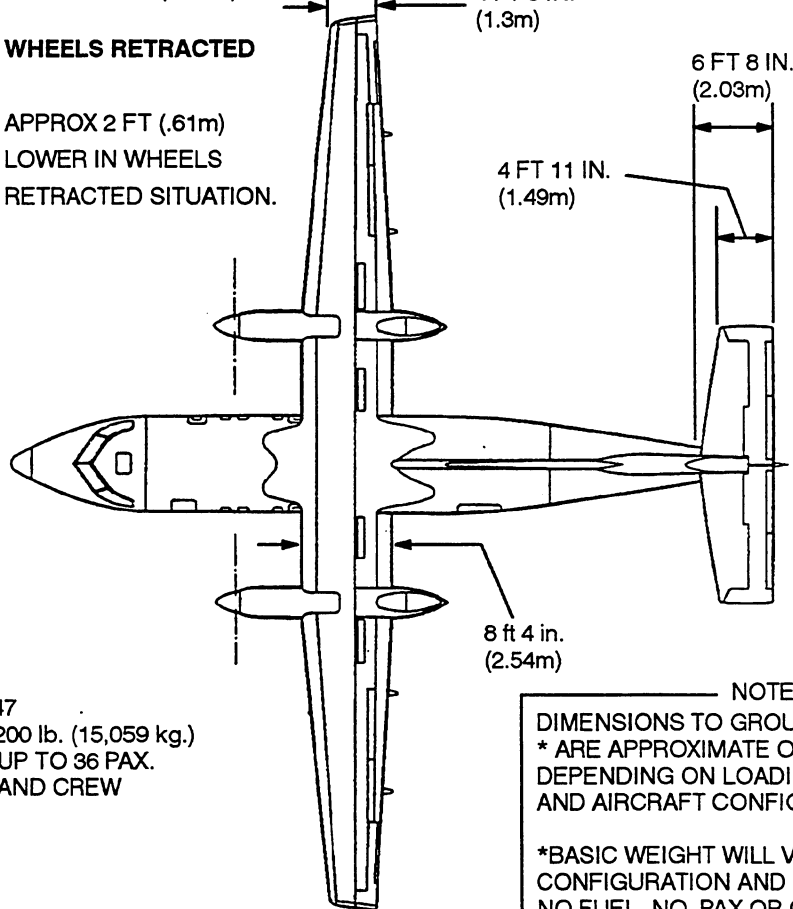
APPROX 2 FT (.61m)

B=43.00IN(1.09m)

LOWER IN WHEELS

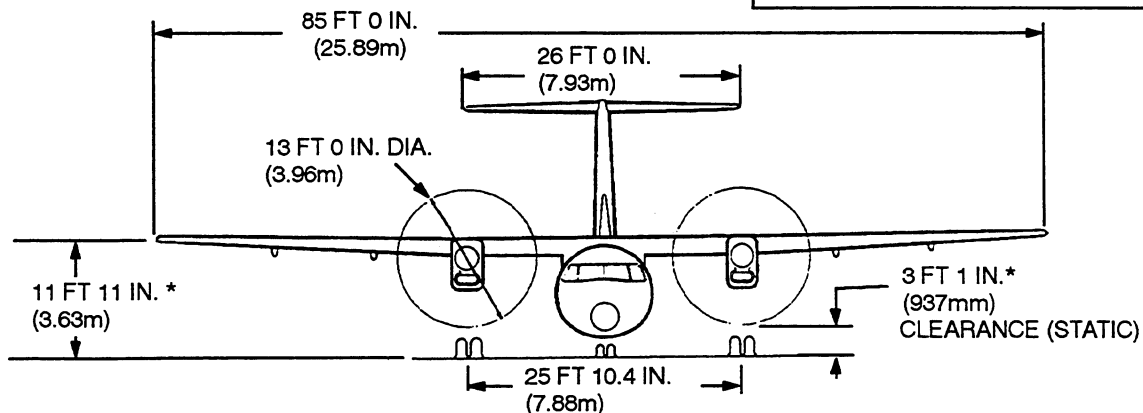
C=63.00IN(1.6m)

RETRACTED SITUATION.



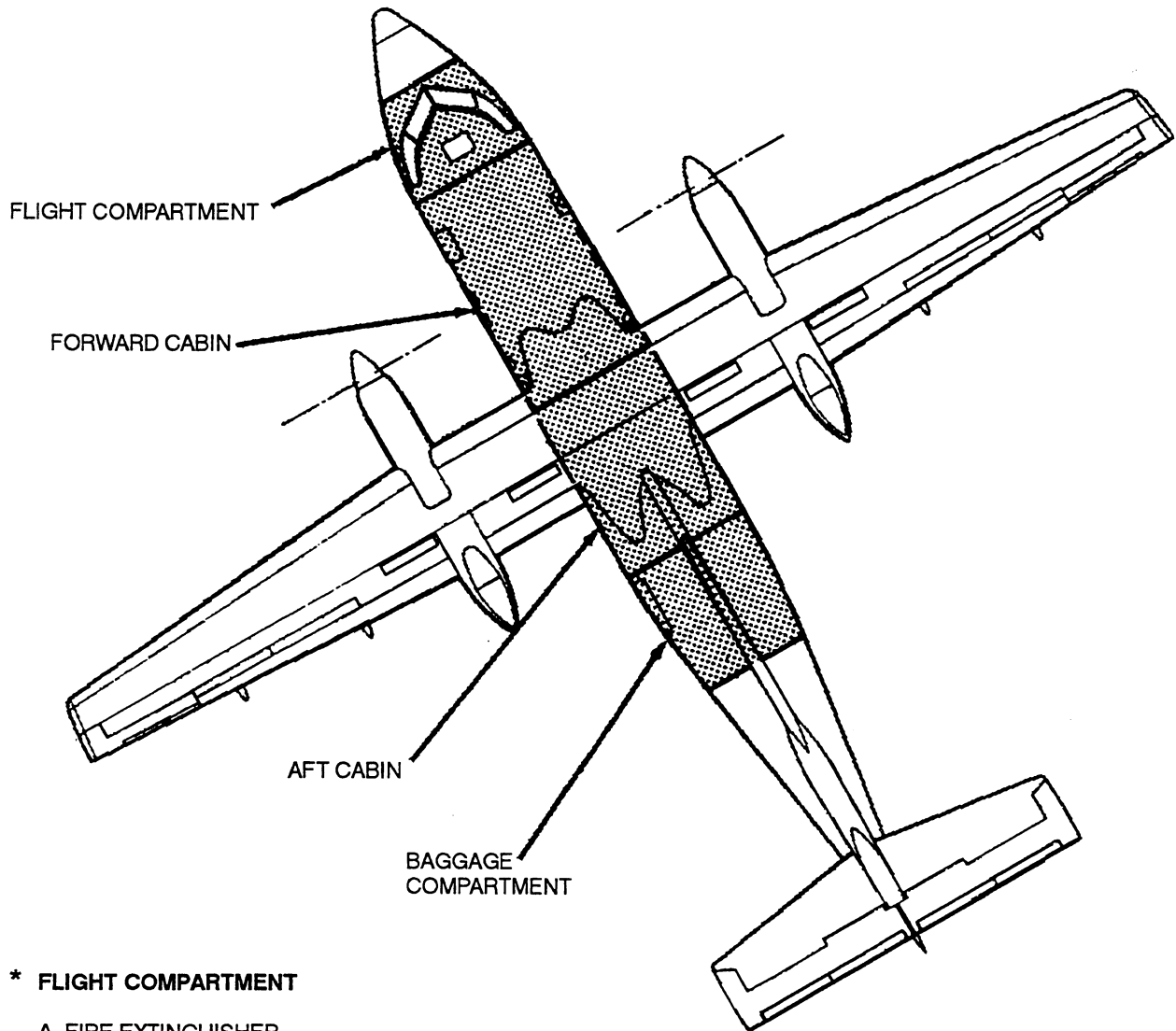
\* BASIC WEIGHT = 21,047  
 GROSS WEIGHT = 33,200 lb. (15,059 kg.)  
 SEATING CAPACITY = UP TO 36 PAX.  
 AND CREW

**NOTE:**  
 DIMENSIONS TO GROUND LINE INDICATED  
 \* ARE APPROXIMATE ONLY AND VARY  
 DEPENDING ON LOADING CONDITIONS  
 AND AIRCRAFT CONFIGURATION.  
 \*BASIC WEIGHT WILL VARY DUE TO  
 CONFIGURATION AND MODIFICATION  
 NO FUEL, NO PAX OR CREW.



**GENERAL ARRANGEMENT**

Boeing Canada  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**



**\* FLIGHT COMPARTMENT**

- A. FIRE EXTINGUISHER
- B. FIRE AXE LOCATION
- C. PORTABLE OXYGEN BOTTLE

**\* FORWARD CABIN**

- A. OXYGEN BOTTLES
- B. GALLEY LOCATION
- C. EMERGENCY DOOR-TYPE II
- D. FIRST AID KIT

**\* AFT CABIN**

- A. EMERGENCY DOORS-TYPE III
- B. FIRE EXTINGUISHER BOTTLES

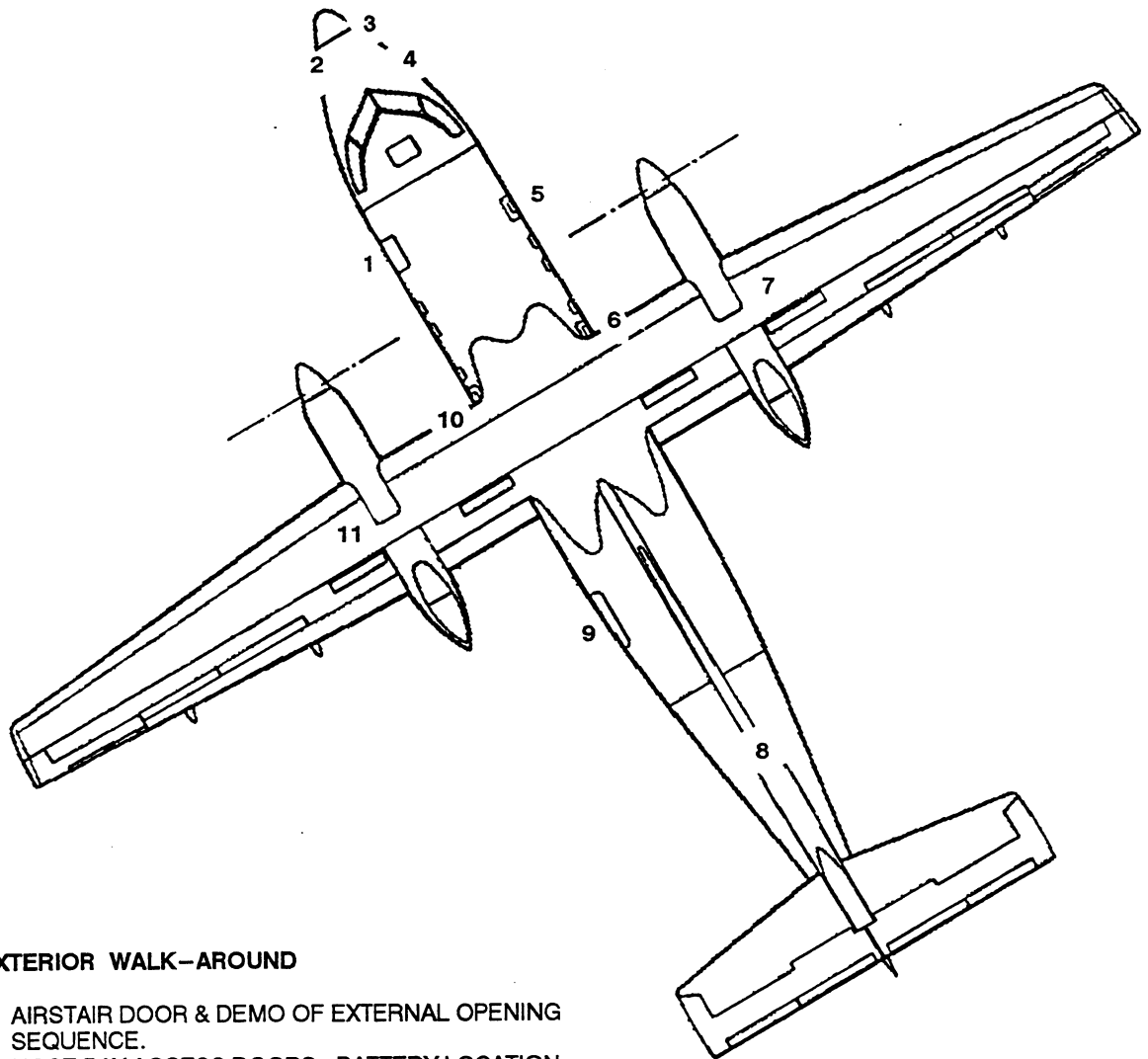
**\* BAGGAGE COMPARTMENT**

- A. ACCESS TO BAGGAGE COMPARTMENT
- B. SMOKE DETECTOR

**FAMILIARIZATION AND LOCATION GUIDE**

Boeing Canada  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

PSM 1-8-14



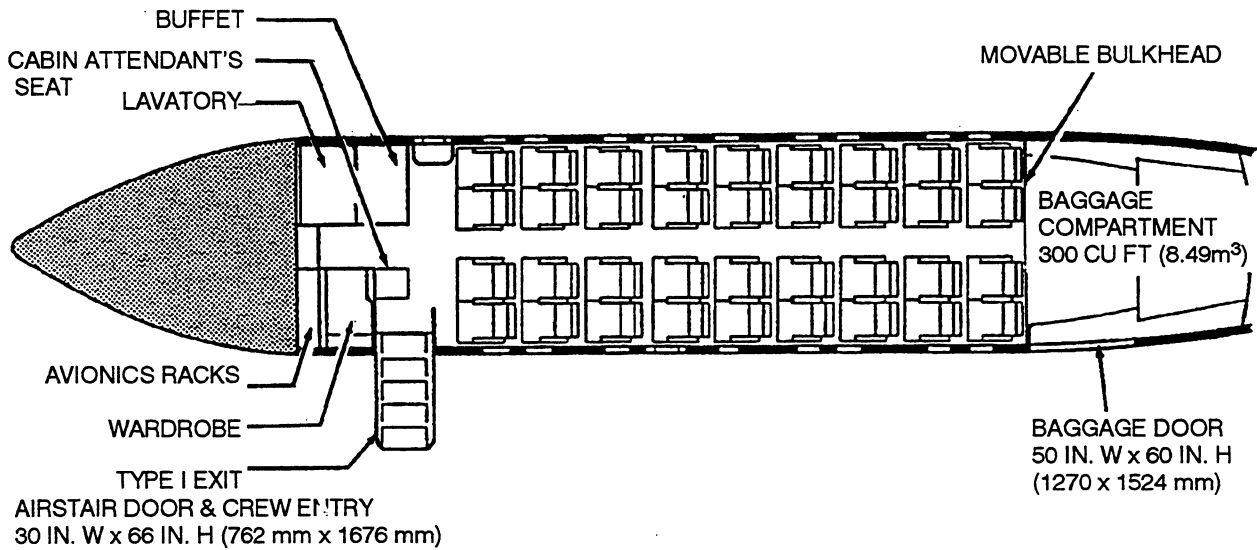
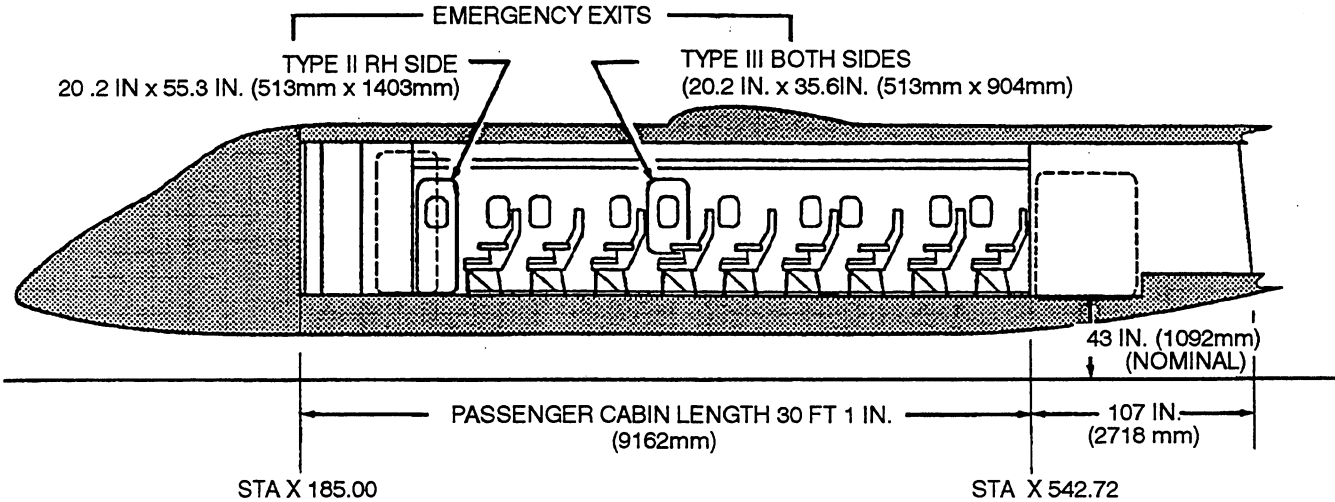
**EXTERIOR WALK-AROUND**

1. AIRSTAIR DOOR & DEMO OF EXTERNAL OPENING SEQUENCE.
2. NOSE BAY ACCESS DOORS-BATTERY LOCATION & MAIN ELECTRICAL CONTACTOR BOX.
3. OXYGEN BOTTLE.
4. NOSE GEAR WHEEL WELL, NOSE GEAR HYDRAULIC & RESERVOIR.
5. TYPE II EMERGENCY DOOR - EXTERNAL OPENING SEQUENCE.
7. TYPE III EMERGENCY DOOR - EXTERNAL OPENING SEQUENCE.
8. REAR COMPARTMENT ACCESS DOOR - OPENING SEQUENCE & FLIGHT DATA RECORDER & COCKPIT RECORDER LOCATION.
9. BAGGAGE COMPARTMENT DOOR - OPENING SEQUENCE.
10. TYPE III EMERGENCY DOOR - EXTERNAL OPENING SEQUENCE.
11. NO.1. NACELLE, MAIN LANDING GEAR & HYDRAULIC RESERVOIR.

**EXTERIOR WALK-AROUND**

# Boeing Canada de Havilland Division DASH 8 CRASH-FIRE-RESCUE INFORMATION

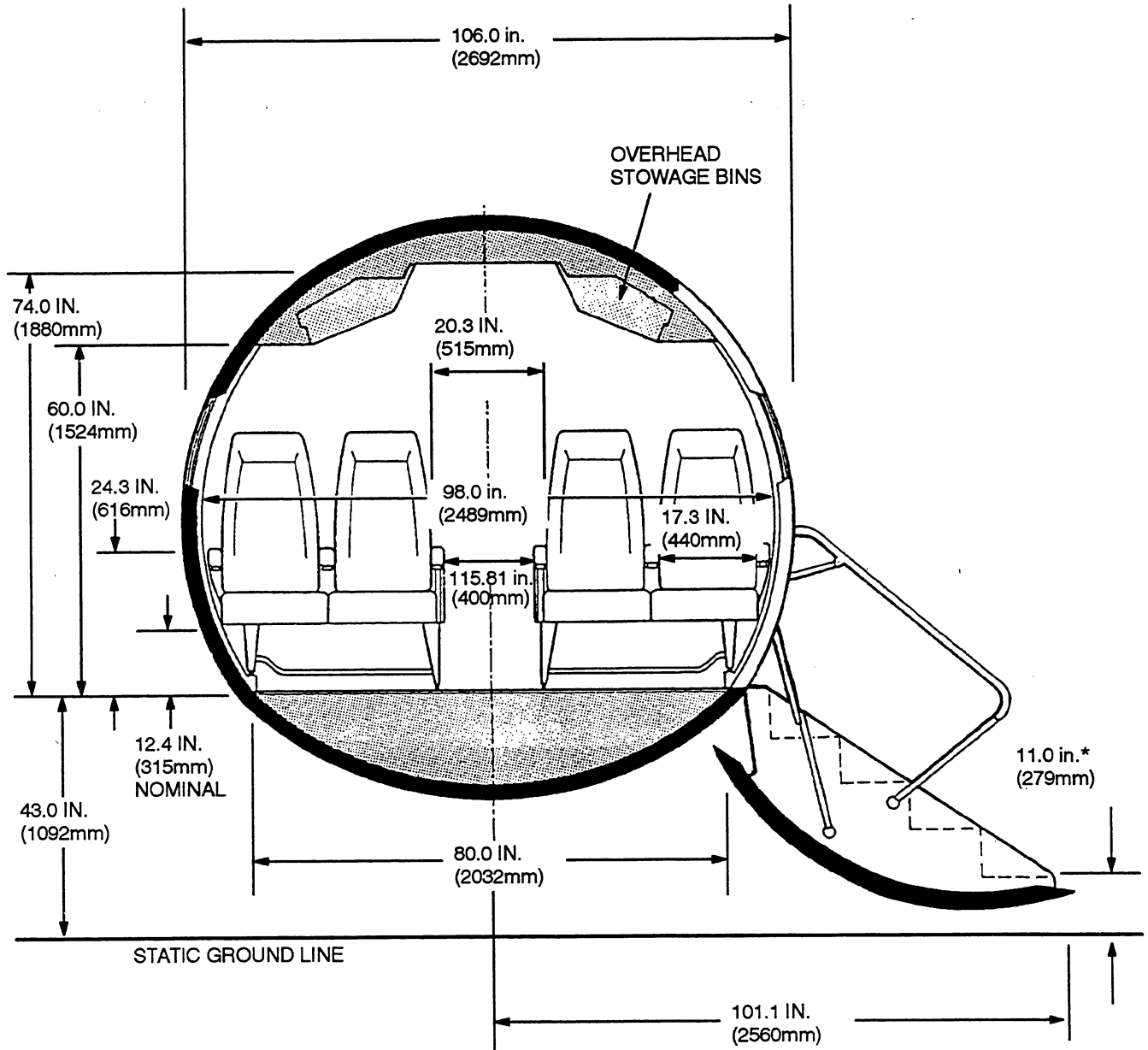
## 36 PASSENGER STANDARD AIRCRAFT



- 36 PASSENGERS AT 31 INCH PITCH (787 mm)
- 8.3 CU FT. BAGGAGE PER PASSENGER (0.24 m<sup>3</sup>)

### INTERIOR ARRANGEMENT

**Boeing Canada**  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

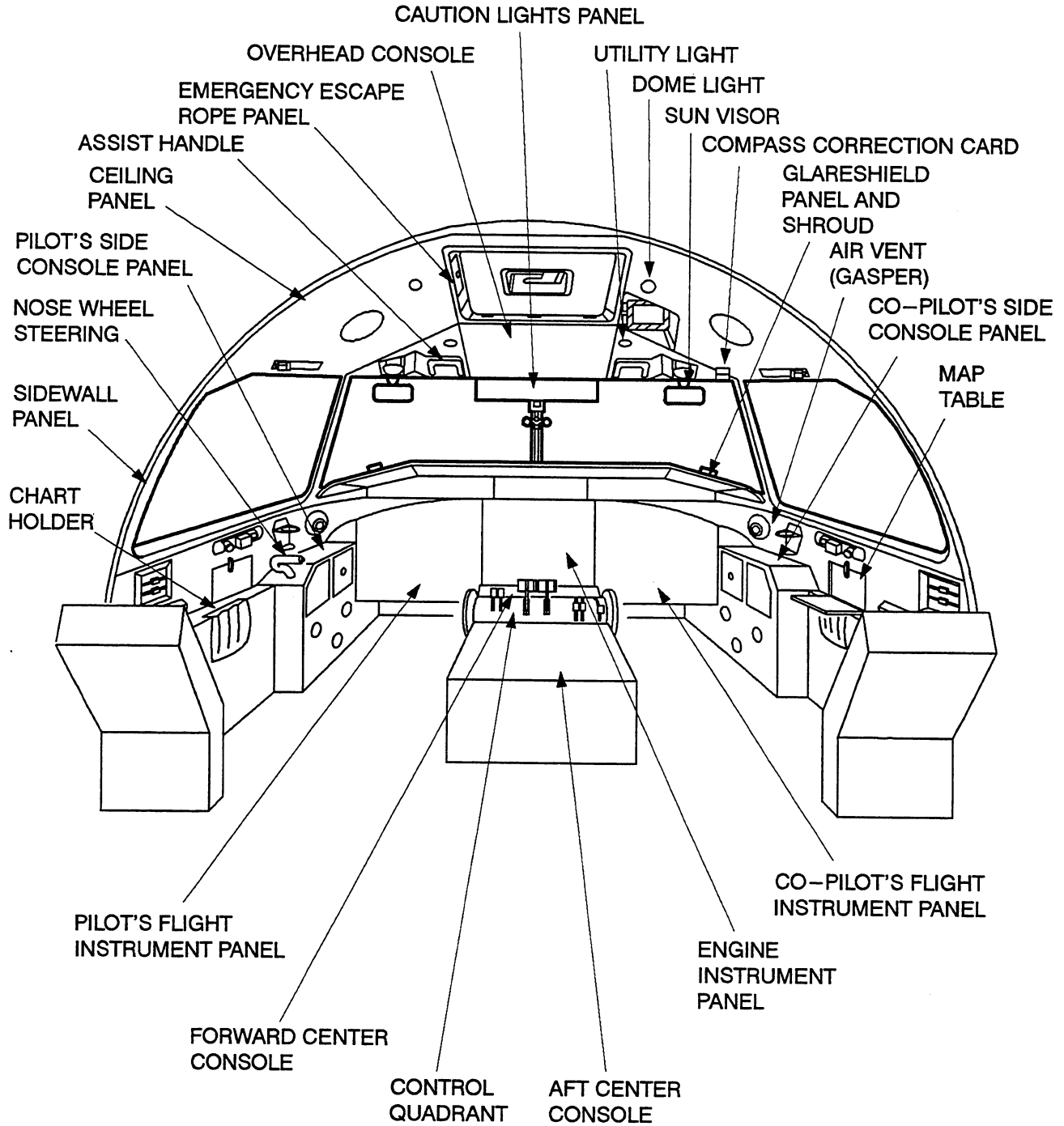


**NOTE:**

\* DIMENSIONS ARE APPROXIMATE AND MAY VARY DEPENDING ON AIRCRAFT CONFIGURATION AND LOADING CONDITIONS.

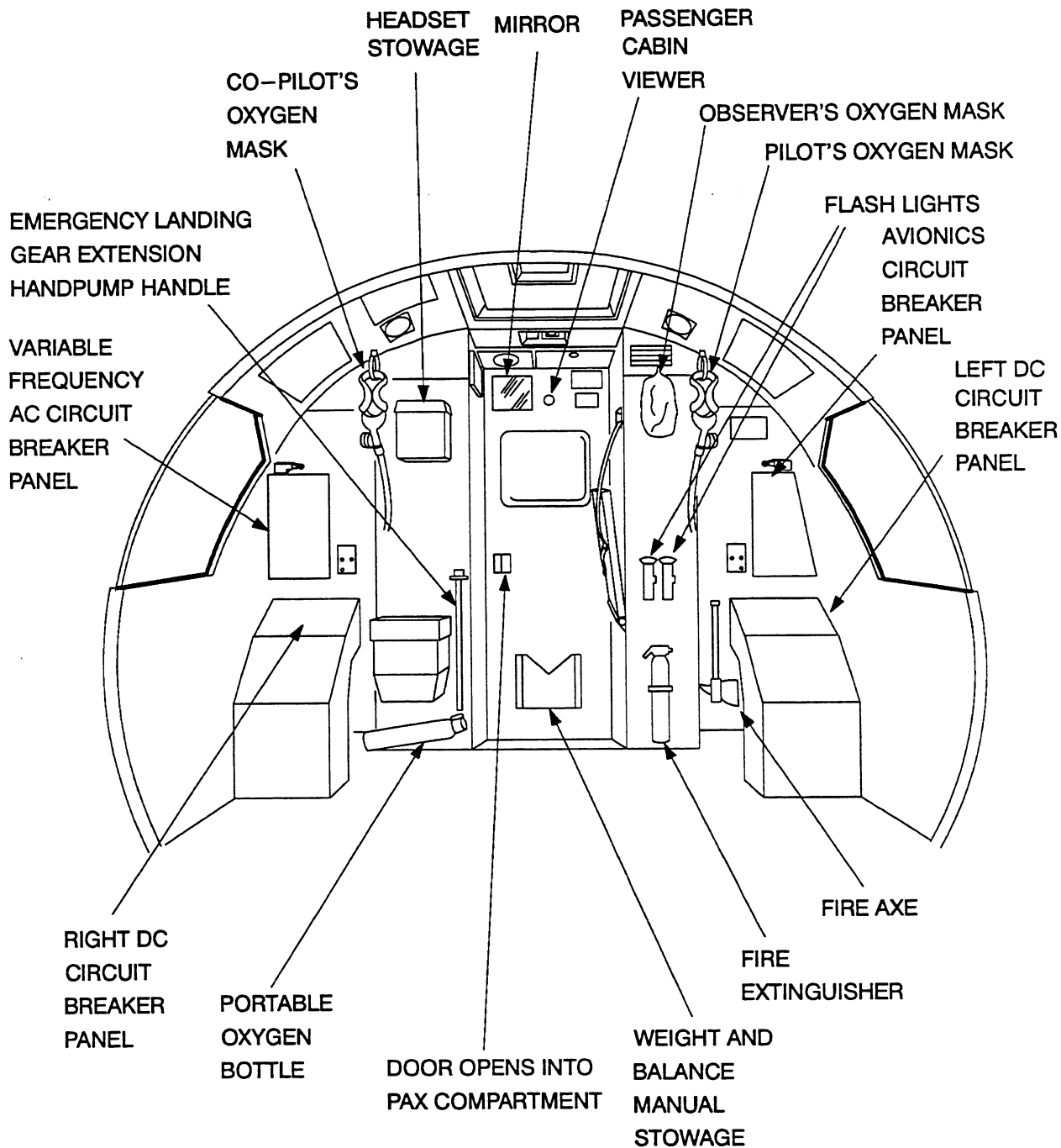
**CABIN CROSS-SECTION**

# Boeing Canada de Havilland Division DASH 8 CRASH-FIRE-RESCUE INFORMATION



**FLIGHT COMPARTMENT (VIEW FORWARD)**

**Boeing Canada**  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**



**FLIGHT COMPARTMENT (VIEW AFT)**

**Boeing Canada**  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

**EXITS****GENERAL**

There are five emergency exits located on the aircraft. A Flight Compartment Emergency Escape Hatch, available to the flight crew, is located in the Flight Compartment roof and is operated by an internal handle. An Airstair door, located on the forward left side of the fuselage, is operated by internal or external handles. The Airstair door incorporates an inflatable seal fed from the 18 psi deicing system. A Type II emergency exit door is located on the right side of the fuselage, opposite the airstair door. Two Type III emergency exit doors are located one on each side of the fuselage, just forward of the wing. The Type II and Type III emergency exit doors each incorporate a window and may be opened by either internal or external handles located below the window. The Type II and Type III emergency exit doors incorporate a compression seal around the outside of the door to contain aircraft pressurization when the doors are closed.

**TYPE II AND TYPE III EMERGENCY EXIT DOOR OPERATION**

The external handle, located below the window, is flush with the door skin and incorporates a push-button for quick-release, enabling the handle to be rotated. Rotating the handle actuates the locking pin and vent dish by a system of pulleys, a cable and a shaft quadrant. A cable guard is installed over the shaft quadrant.

To remove either the Type II or Type III emergency exit door using the external handle, push the quick-release button to release the handle. Turn the handle counterclockwise to open the vent and retract the locking pin. Push the door inward.

**AIRSTAIR DOOR OPERATION**

The Airstair door is opened externally by operation of the door handle lever located on the left side of the fuselage just forward of the door. Initial movement of the handle trips the door seal pressurizing valve to release the seal pressure allowing cabin pressure to deplete. Continued movement of the handle moves the door upward and inward to clear the ten pressure pads from their mating stops so that the door may be manually pulled open. Door lowering is assisted by a door counter-balance system.

**SERVICE DOORS****BAGGAGE DOOR OPERATION**

The Baggage door is located on the left side of the rear fuselage. The door is opened and closed manually using an external handle which normally is flush with the door skin. A quick-release button is located in the center of the handle.

To open the Baggage door, release the handle from the stowed position by pushing the quick-release button. Rotate the handle 180 degrees counterclockwise to unlock the door and initiate an inward and upward movement. Stow the handle by pressing it back into its recess in the door and, while supporting the door, manually raise to the fully open position. Secure the door in the open position by engaging the door support strut.

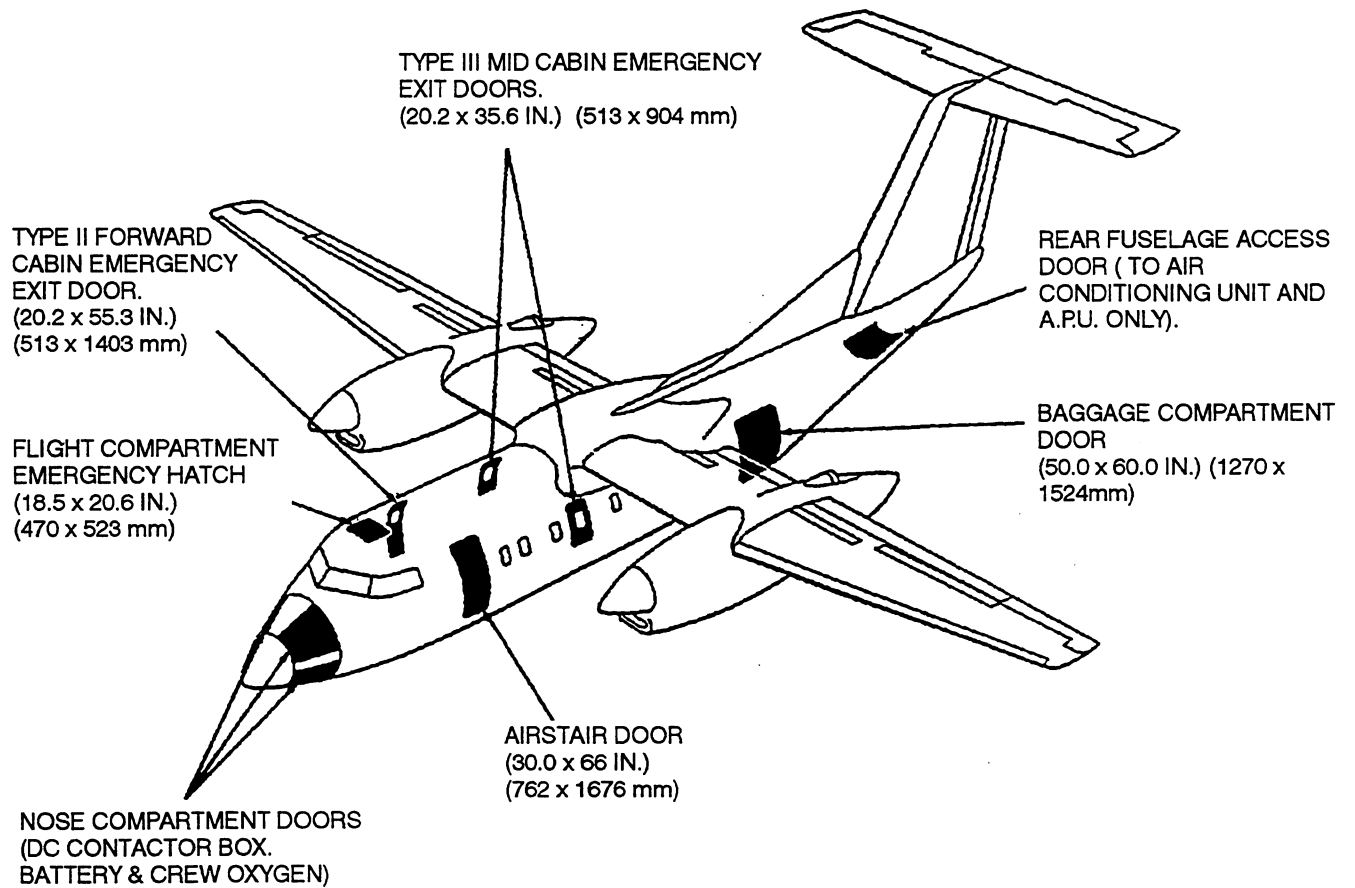
**NOTE**

Cabin compartment emergency entry from the baggage compartment is not normally possible.



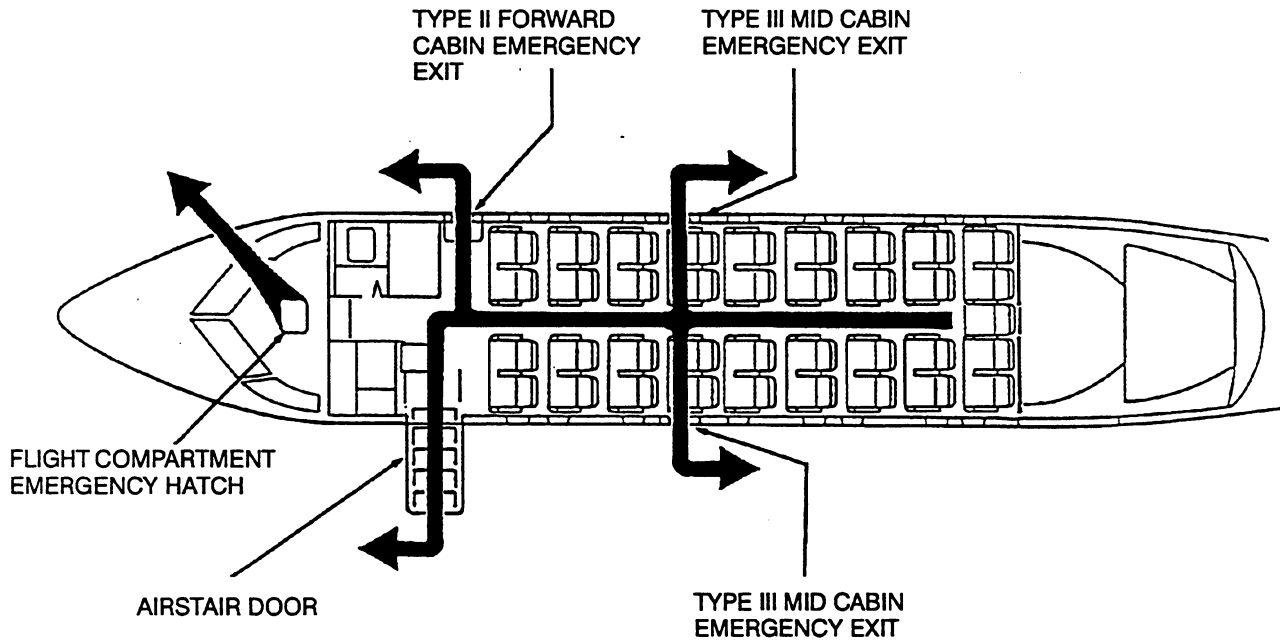
Boeing Canada  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

PSM 1-8-14



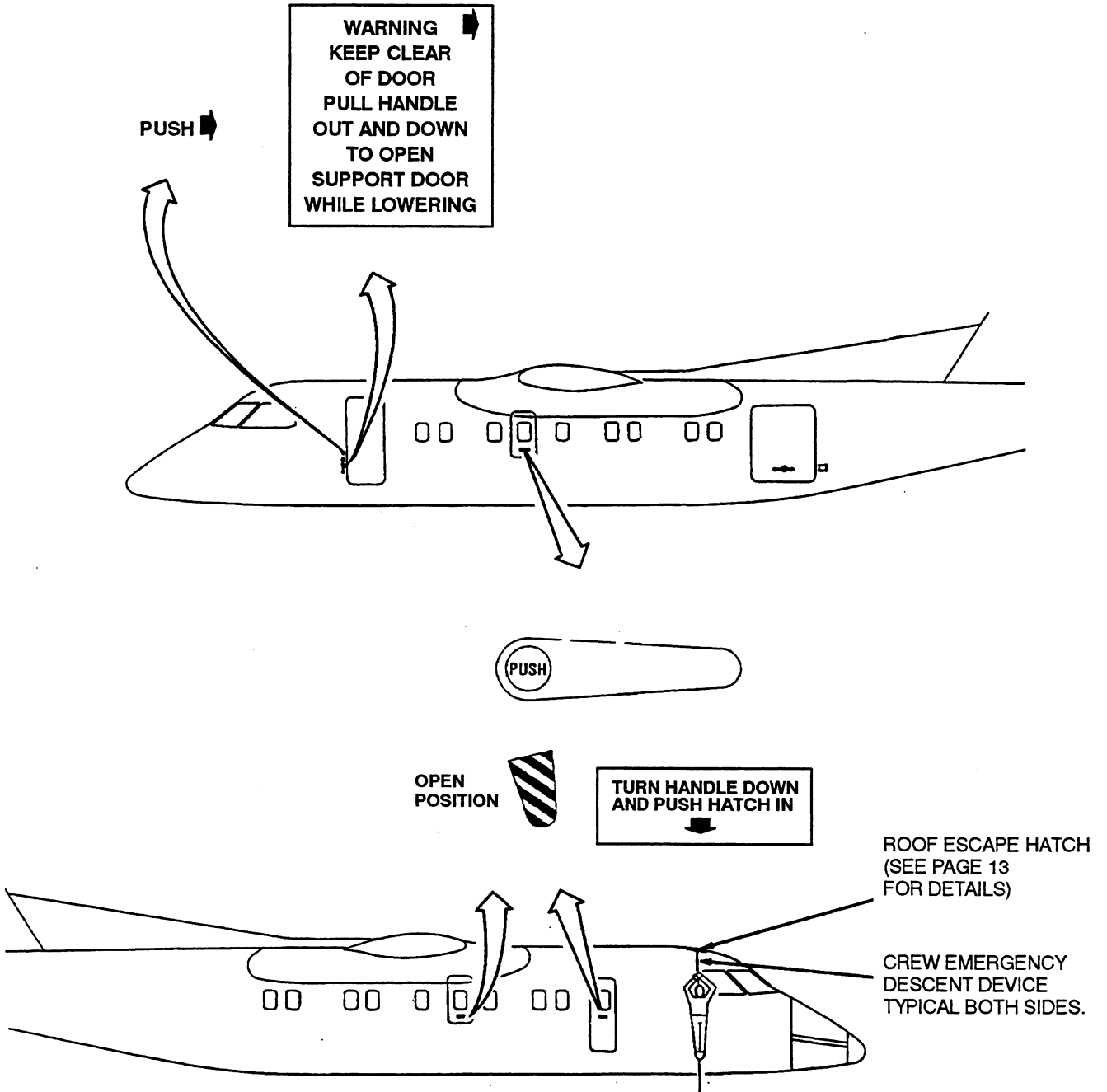
**AIRCRAFT DOORS AND GROUND SERVICE PANELS**

**Boeing Canada**  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**



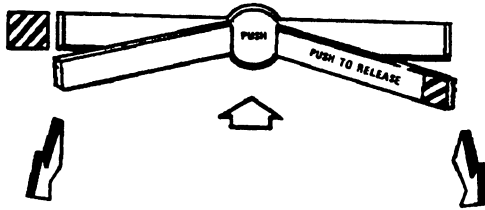
**EVACUATION ROUTES**

Boeing Canada  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

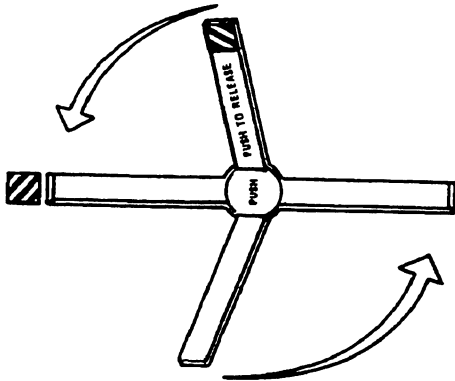


**PASSENGER AND CREW ESCAPE SYSTEMS**

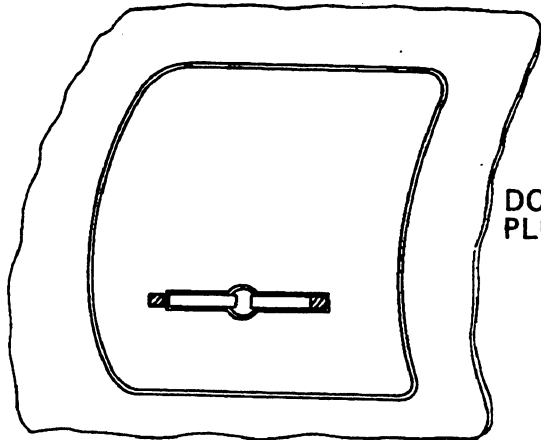
**Boeing Canada**  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**



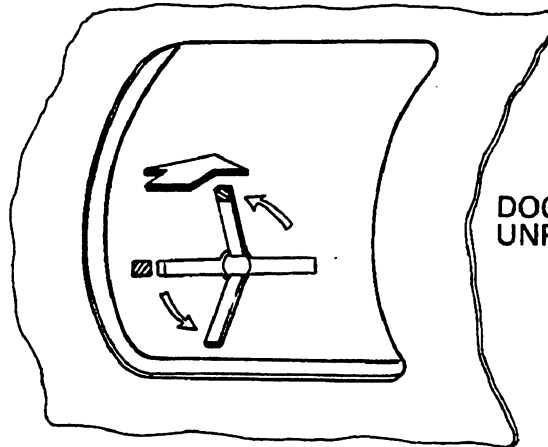
**1 TO OPEN, PUSH ON BUTTON  
DOOR HANDLES SPRING  
OUTWARDS**



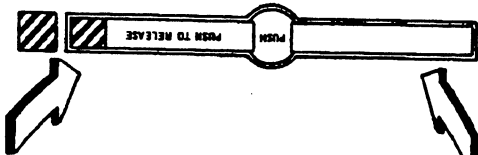
**2 TURN HANDLES COUNTER  
CLOCKWISE 180°  
(DOOR MOVES INWARDS)**



**DOOR  
PLUGGED**



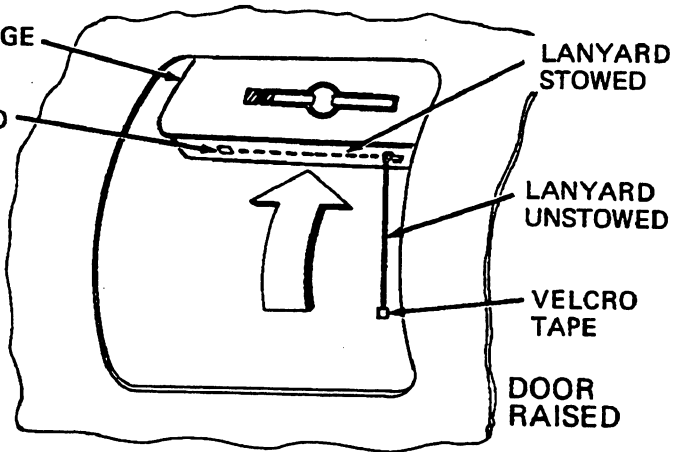
**DOOR  
UNPLUGGED**



**3 PUSH HANDLES IN FLUSH  
WITH FUSELAGE, THEN  
PUSH DOOR UPWARDS TO  
OPEN**

**4 TO CLOSE, PULL BAGGAGE DOOR  
DOWN AND TURN HANDLE CLOCK-  
WISE 180°**

**BAGGAGE  
DOOR  
VELCRO  
PAD**



**LANYARD  
STOWED**

**LANYARD  
UNSTOWED**

**VELCRO  
TAPE**

**DOOR  
RAISED**

**WARNING**

LIFT DOOR TO OPEN  
HANDLE TO BE FLUSH BEFORE LIFTING  
WHEN RAISING OR LOWERING KEEP CLEAR OF  
DOOR PATH

**NOTE:**  
UNSTOW LANYARD AS REQUIRED FOR CLOSING DOOR.  
ONCE DOOR IS PULLED DOWN, RESTOW LANYARD  
BEFORE REPLUGGING DOOR (PRE-MOD 8/1056  
AIRPLANES). ON AIRPLANES INCORPORATING MOD  
8/1056, THE LANYARD STRAP IS REPLACED BY A  
SUPPORT STRUT. SECURE THE STRUT TO THE RIGHT  
SIDE DOOR FRAME BRACKET TO SUPPORT THE DOOR  
IN THE OPEN POSITION. RESTOW SUPPORT STRUT  
INTO SPRING CLIP BEFORE RE-PLUGGING DOOR.

**BAGGAGE DOOR OPERATION**

**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

**FLIGHT COMPARTMENT EMERGENCY ESCAPE HATCH**

**DESCRIPTION**

The Flight Compartment escape hatch, located in the Flight Compartment roof, is completely detachable for emergency exit or can be partially opened for ventilation when the aircraft is on the ground. The hatch is mounted at the rear on two support fittings and at the front by two locking and release fittings. An operating handle, located in the center of the hatch, is retained in an open or closed position by an overcenter spring. The handle operates a transversely-mounted torque shaft assembly with arms attached at each end. Rollers at the end of each arm engage detented locking release fittings installed in the Flight Compartment roof structure.

A seal is installed around the edge of the hatch to contain the aircraft pressurization when the hatch is closed.

**OPERATION**

To open the Flight Compartment escape hatch, rotate the handle 72 degrees counterclockwise. A mechanical linkage connected to the handle rotates the torque tube and the rollers move forward in the fittings where they are supported by the spring-loaded detents. Controlled by the geometry of the torque tube and the rollers, the hatch pivots about the rear support fittings and opens approximately one inch at the front. Opening the hatch permits depressurization and provides a modest amount of ventilation to the Flight Compartment.

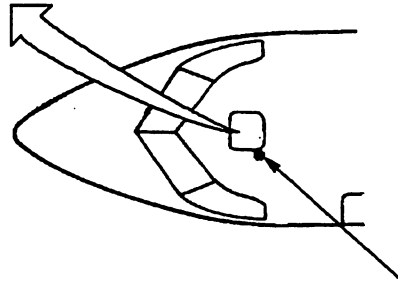
A downward pull on the handle of approximately 40 pounds releases the rollers against the action of the forward locking and release detent springs. The hatch may then be completely removed.

# Boeing Canada de Havilland Division DASH 8 CRASH-FIRE-RESCUE INFORMATION

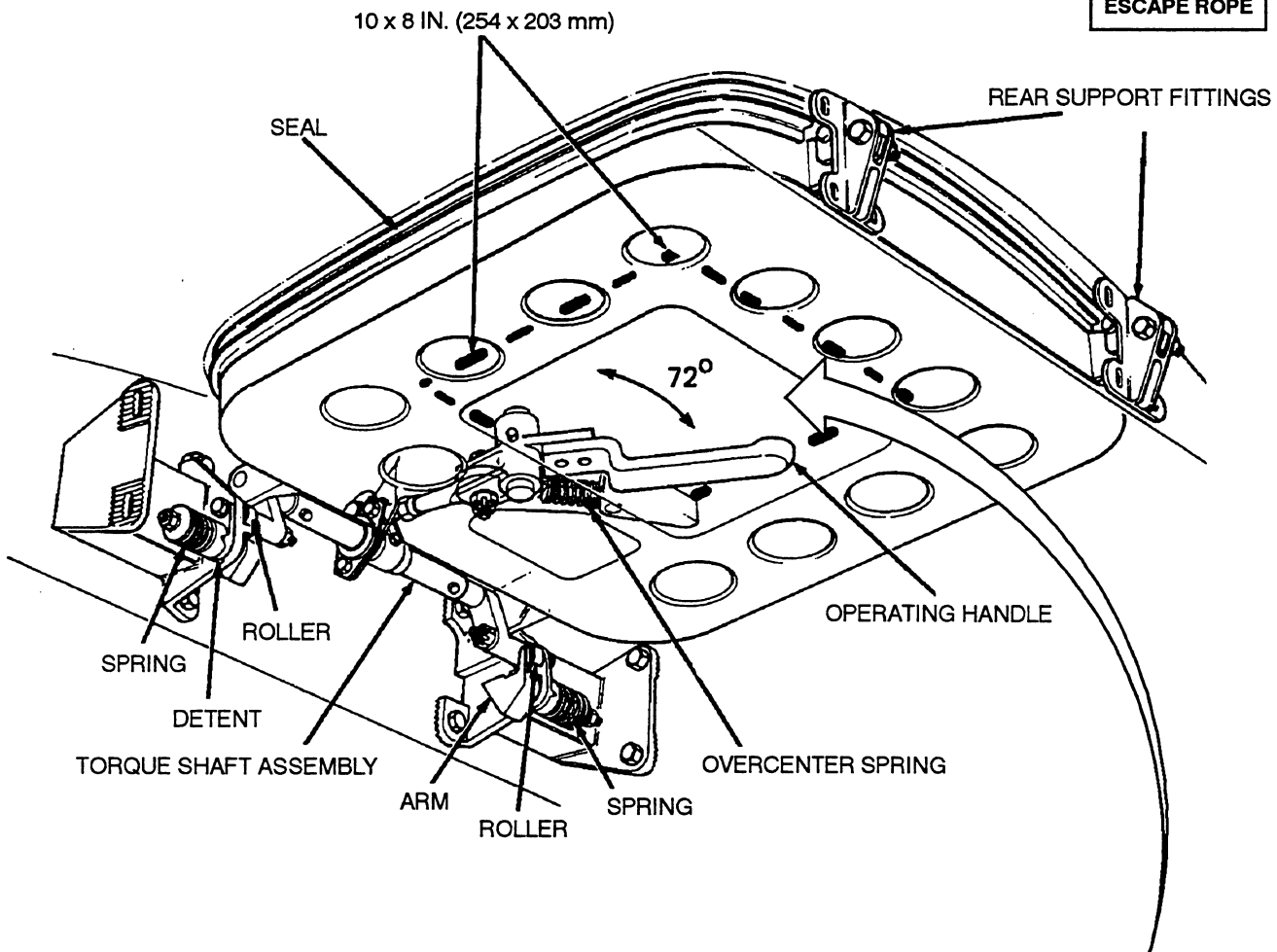
FLIGHT COMPARTMENT  
EMERGENCY HATCH  
18.5" X 20.6" (470 X 523 mm)

**NOTE:**

IN AN EMERGENCY IT MAY BE POSSIBLE BY CUTTING THROUGH THE OUTER SKIN TO GAIN ACCESS TO REPOSITION THE OPERATING HANDLE FROM OUTSIDE AND THEN FORCE THE HATCH DOWNWARDS.



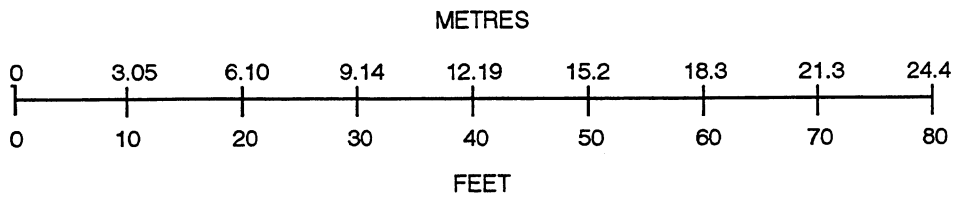
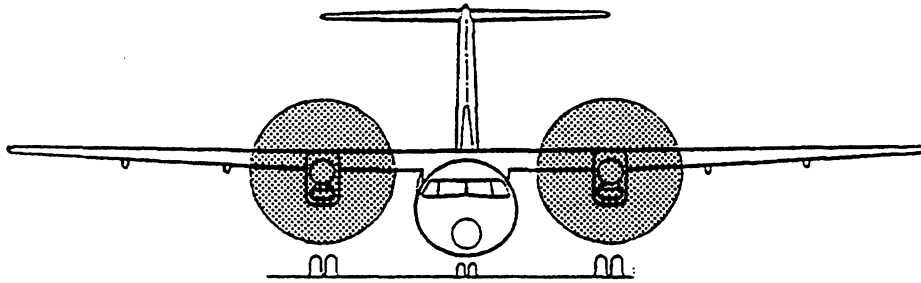
**EMERGENCY  
ESCAPE ROPE**



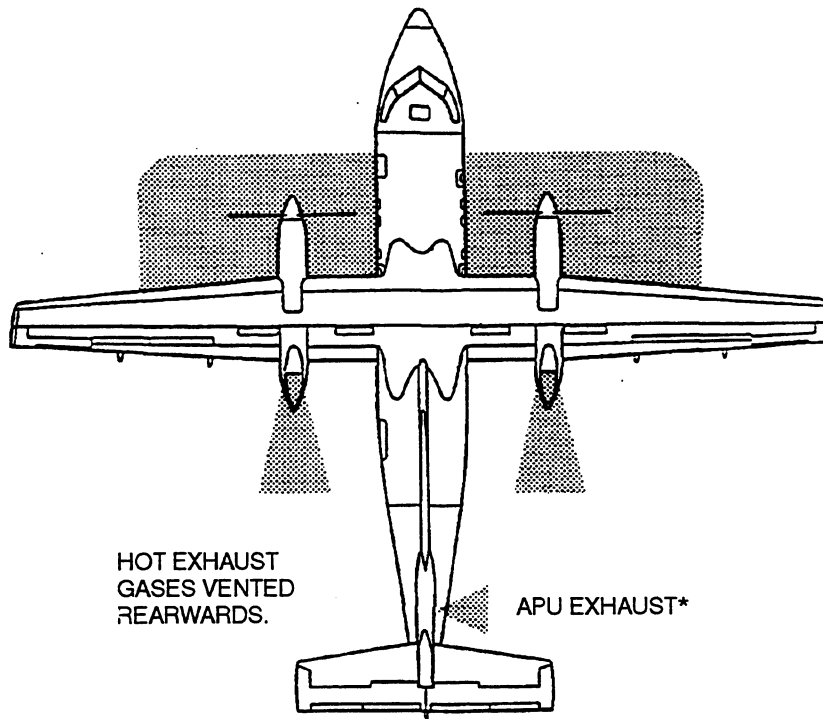
## FLIGHT COMPARTMENT EMERGENCY ESCAPE HATCH

Boeing Canada  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

PSM 1-8-14

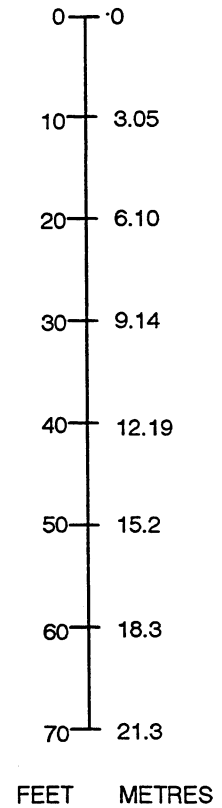


 PERSONNEL DANGER AREAS  
(WHEN ENGINES ARE OPERATING)



HOT EXHAUST  
GASES VENTED  
REARWARDS.

APU EXHAUST\*



\*NOT ALL AIRCRAFT

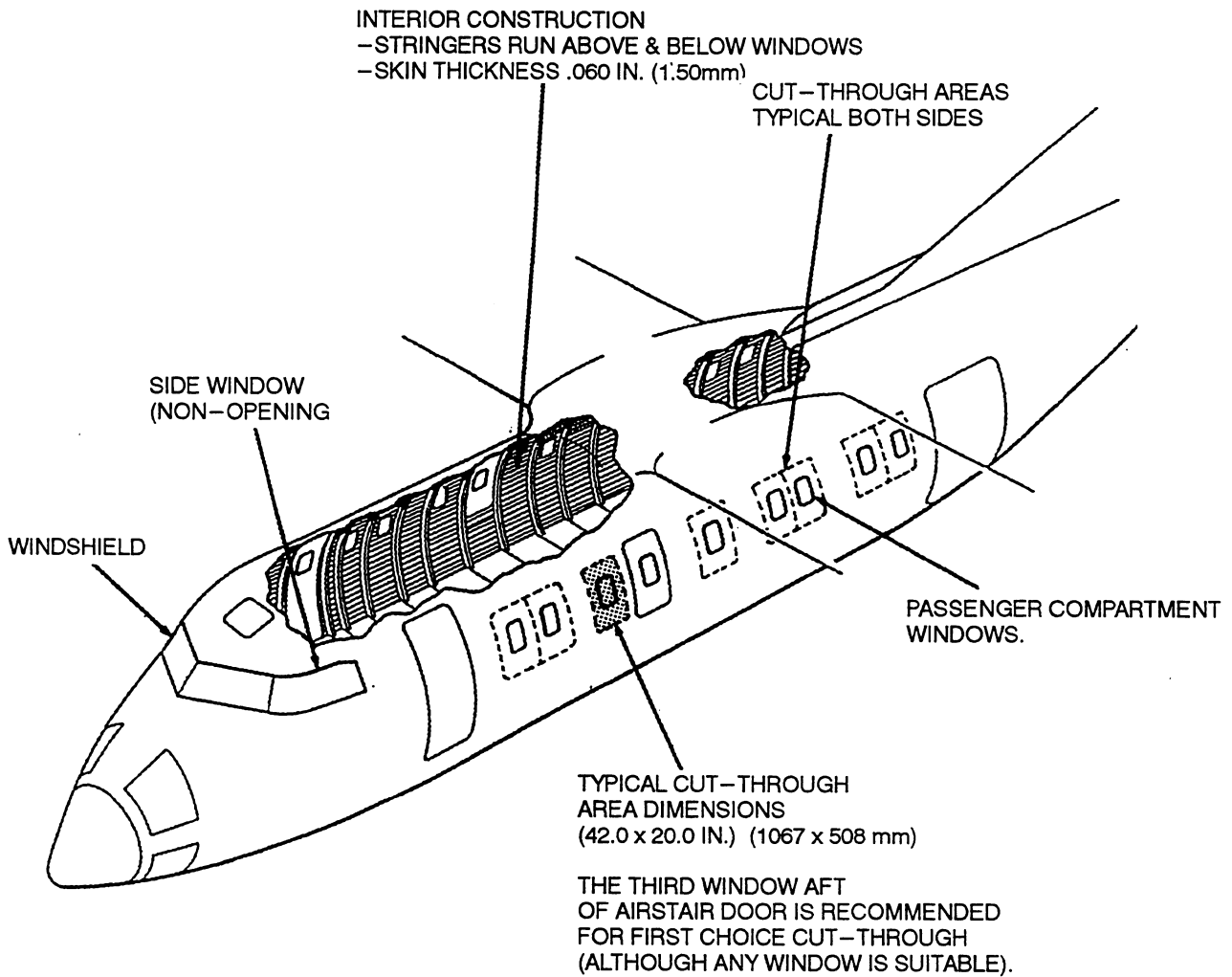
**ENGINE DANGER AREAS**

# Boeing Canada

de Havilland Division

## DASH 8 CRASH-FIRE-RESCUE INFORMATION

NOTE:  
"CUT-THROUGH" AREAS REQUIRE PORTABLE  
METAL-CUTTING EQUIPMENT. IT IS RECOMMENDED  
THAT MAJOR EFFORT TO GAIN ACCESS BE  
DIRECTED TO HATCHES AND DOORS DUE TO THE  
TYPE OF STRUCTURE AND POSSIBLE INJURY TO  
PERSONNEL WITHIN.

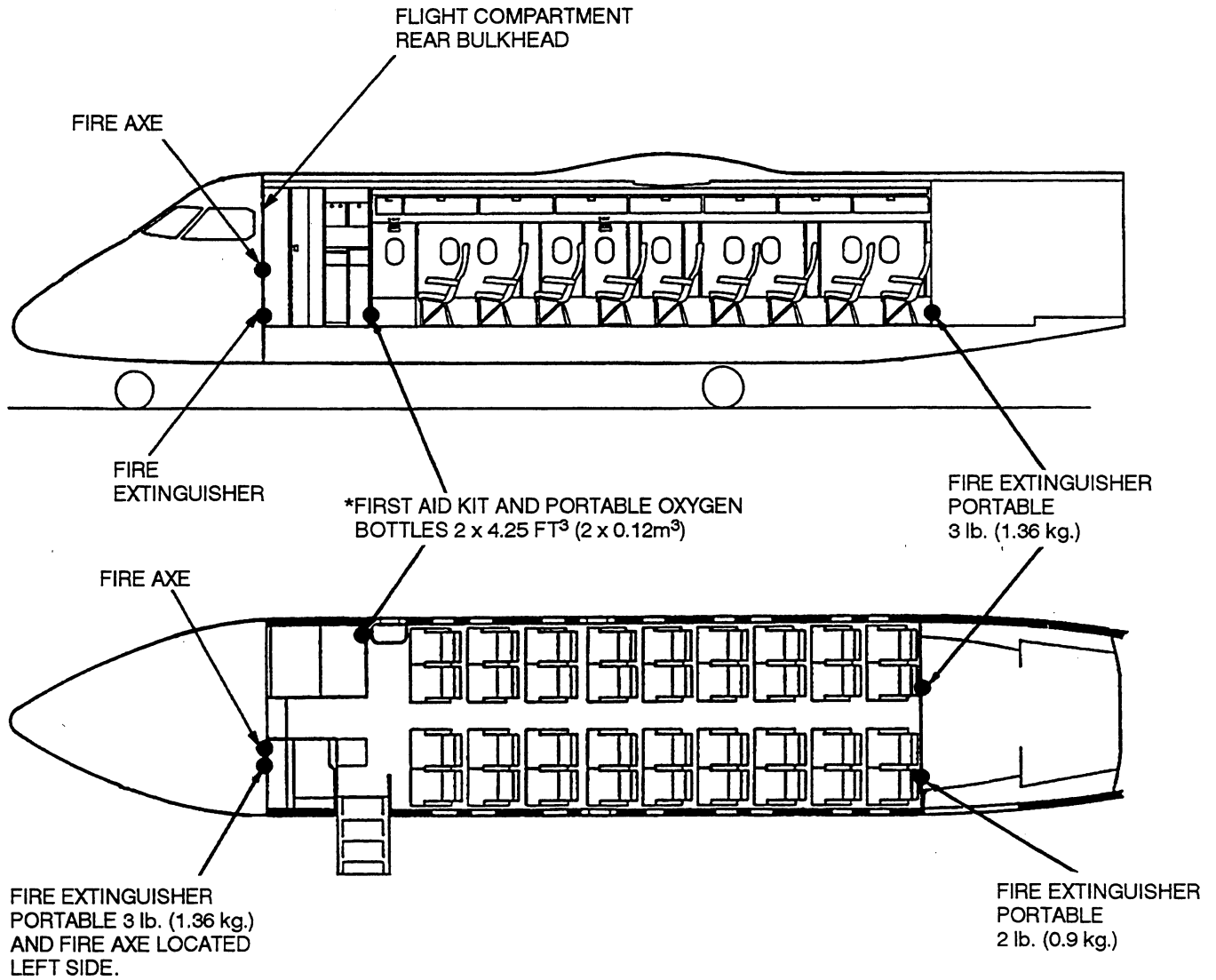


### CUT-THROUGH AREAS



Boeing Canada  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

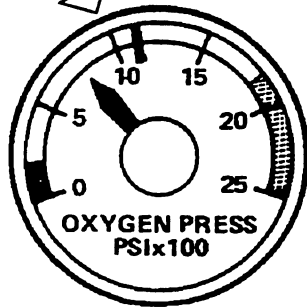
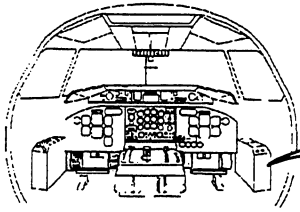
PSM 1-8-14



\*NOTE: MAY VARY WITH AIRLINE & CONFIGURATION

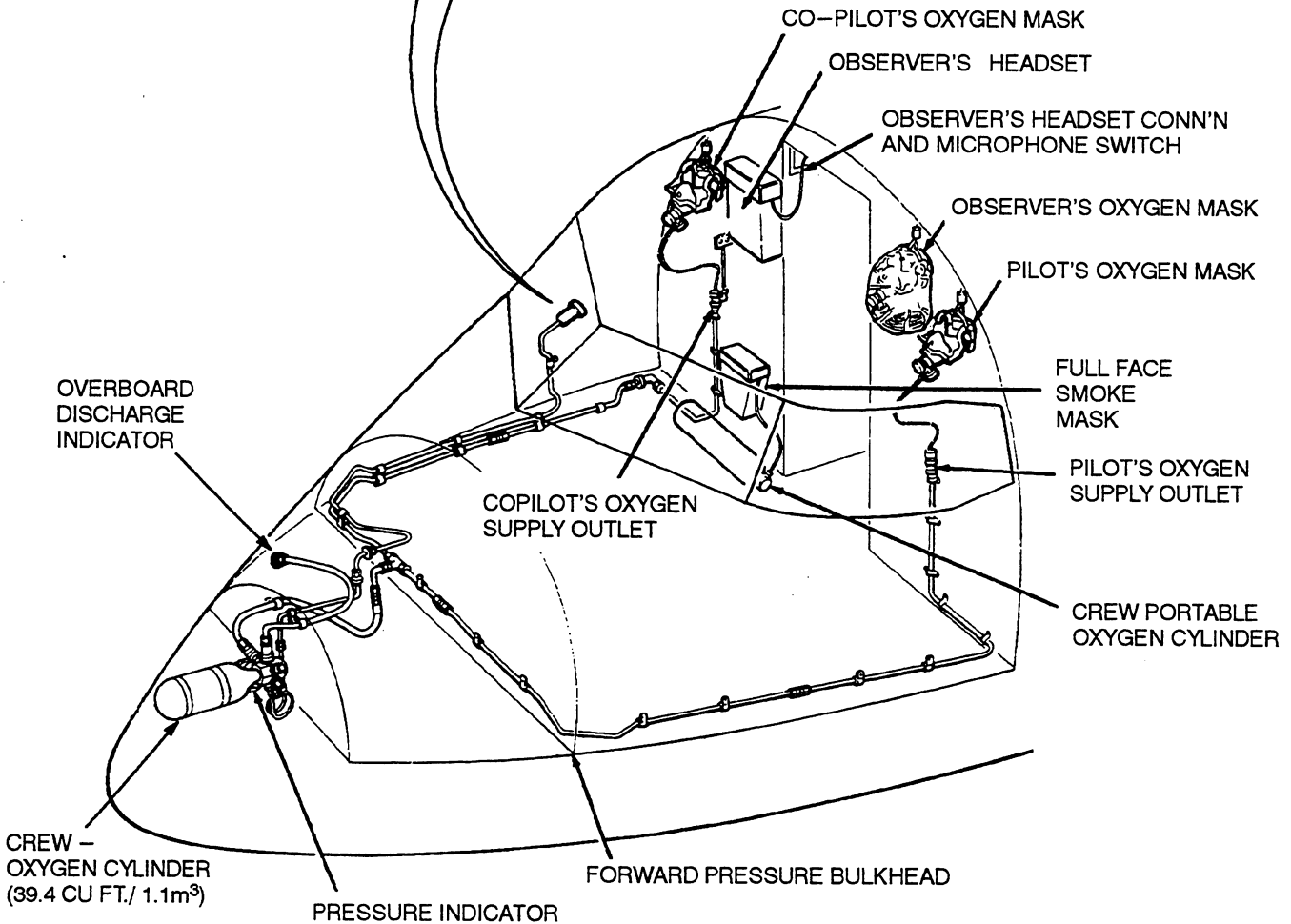
**FUSELAGE SAFETY EQUIPMENT LOCATIONS**

# Boeing Canada de Havilland Division DASH 8 CRASH-FIRE-RESCUE INFORMATION



RANGE MARKS	
RED:	0 - 250 psi (DANGER LEVEL)
GREEN:	250 - 1800 psi (USEABLE RANGE)
YELLOW:	1800 - 2500 psi (OVERPRESSURE)
BLUE RADIAL:	1300 psi (MINIMUM DISPATCH)

OXYGEN INDICATOR



**CREW OXYGEN LOCATIONS**

**DASH 8 CRASH-FIRE-RESCUE INFORMATION****FIRE CONTROL RECOMMENDATIONS**

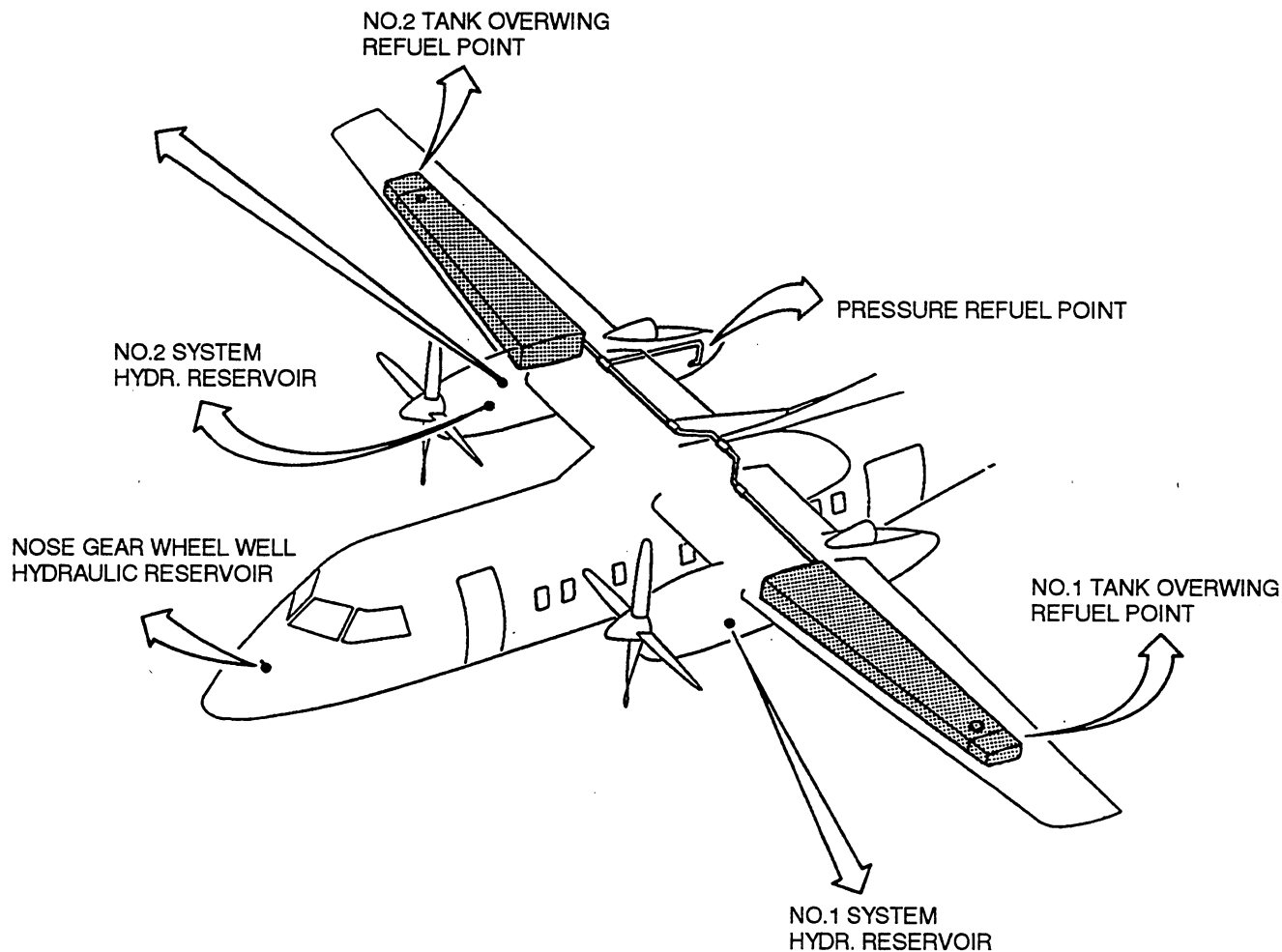
FIRE AREA	EXTINGUISHER TYPE			NOTES
	PREFERRED	ALTERNATIVE	AVOID	
ENGINE FIRES	HALON 1211	FOAM	CO <sub>2</sub> CAN DAMAGE ENGINE. DRY CHEMICAL IS CORROSIVE.	1. WHEELS ARE EQUIPPED WITH FUSIBLE PLUGS WHICH WILL BLOW AT 288°F (142°C). 2. APPROACH LANDING GEAR FROM FORWARD OR AFT. STAND UPWIND OF FIRE TO AVOID 'SKY-DROL' FUMES. ALL WHEELS ARE FORGED ALUMINUM.
FUEL FIRE	1. DRY CHEMICAL POWDER FOR LEAKING FUEL. 2. WATER FOG OR FOAM ON GROUND SPILL AREA.			
WHEEL FIRE	DRY CHEMICAL POWDER	HALON 1211	CO <sub>2</sub> - WHEEL BREAKAGE IS POSSIBLE.	
ELECTRICAL FIRE	HALON 1211	DRY CHEMICAL POWDER/CO <sub>2</sub>	WATER	
HYDRAULIC SERVICE BAY FIRE	HALON 1211	DRY CHEMICAL POWDER/CO <sub>2</sub>	WATER	
ELECTRICAL/ELECTRONIC SERVICE BAY FIRE	HALON 1211	DRY CHEMICAL POWDER/CO <sub>2</sub>	WATER	
GALLEY FIRE	HALON 1211	DRY CHEMICAL POWDER	WATER	
FLIGHT COMPARTMENT FIRE	HALON 1211	DRY CHEMICAL POWDER	WATER	
CABIN COMPARTMENT FIRE	HALON 1211	DRY CHEMICAL POWDER	WATER	
CARGO COMPARTMENT FIRE	HALON 1211	DRY CHEMICAL POWDER/CO <sub>2</sub>	WATER	

# Boeing Canada de Havilland Division

## DASH 8 CRASH-FIRE-RESCUE INFORMATION

TOTAL OIL CAPACITY FOR BOTH ENGINES		
IMP GAL.	US GAL.	LITRES
9.24	11.09	42

HYDRAULIC FLUID			
RESERVOIR	IMP QTS	US QTS	LITRES
NO. 1 SYSTEM	2.20	2.68	2.50
NO. 2 SYSTEM	4.31	5.19	4.90
NOSE	1.06	1.25	1.20
ALL SYSTEMS USE PHOSPHATE ESTER-BASED TYPE IV FLUID eg. SKYDROL			



TOTAL FUEL CAPACITY					
BASED ON JET A-1 S.G. OF 0.816 (SINGLE TANK DIVIDE BY 2)					
	LB	KG	IMP GAL	US GAL	LITRES
STANDARD FUEL TANKS	5765	2614	705	846	3203
EXTENDED RANGE TANKS	10,433	4732	1276	1532	5800

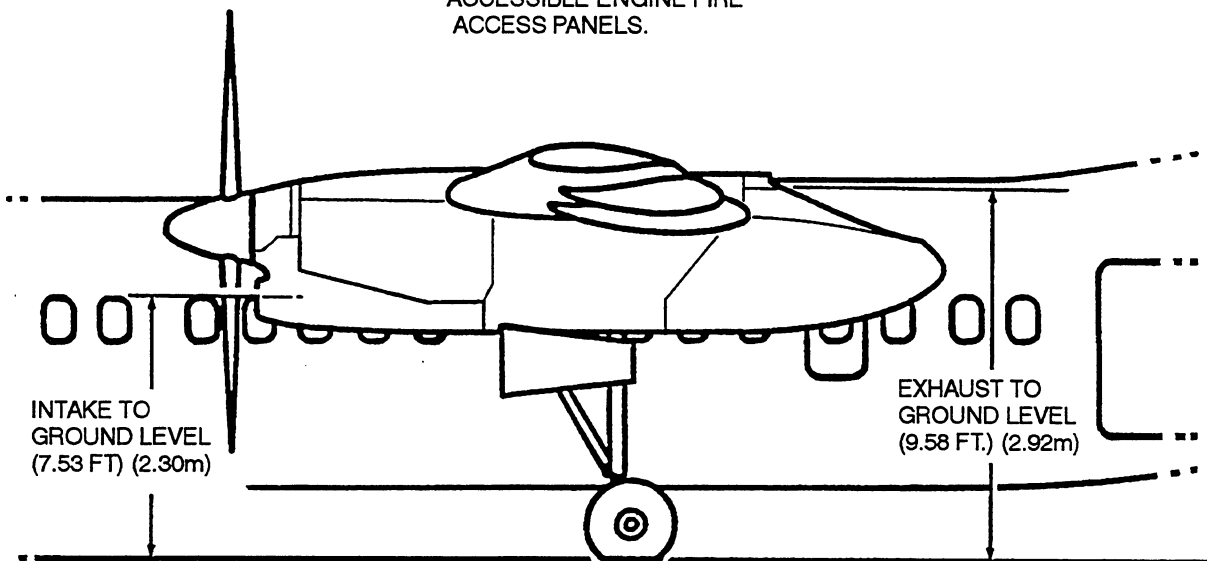
### FLAMMABLE MATERIAL LOCATIONS

Boeing Canada  
de Havilland Division  
**DASH 8 CRASH-FIRE-RESCUE INFORMATION**

PSM 1-8-14

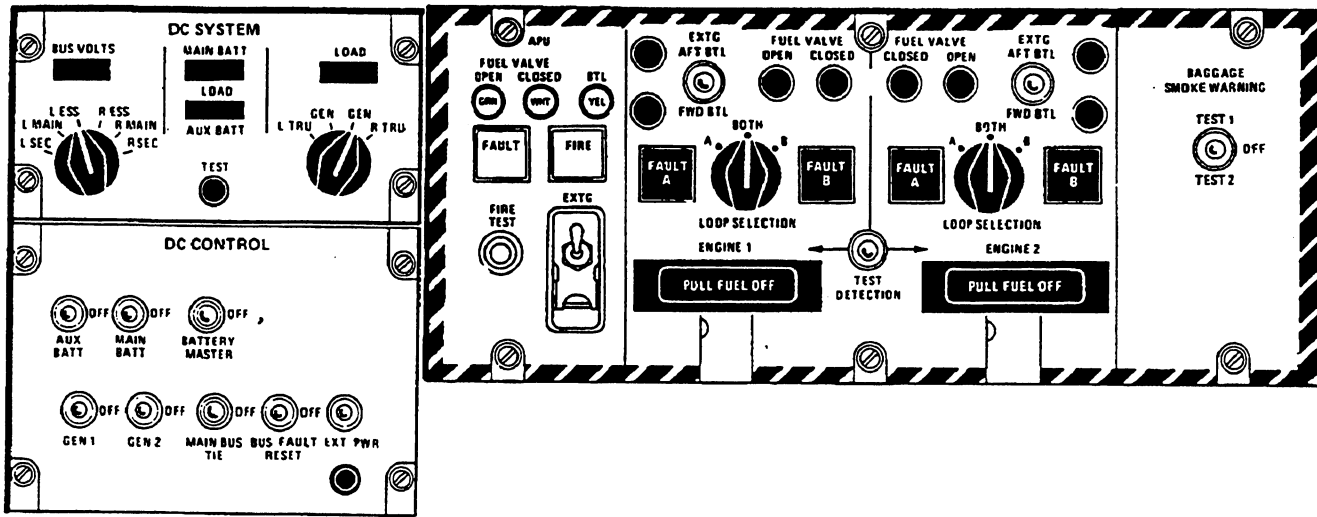
NOTE:

- \* APPROX. 2 FT. LOWER IN WHEELS-UP SITUATION.
- \* THERE ARE NO EXTERNALLY ACCESSIBLE ENGINE FIRE ACCESS PANELS.



**ENGINE FIRE ACCESS LOCATIONS**

# Boeing Canada de Havilland Division DASH 8 CRASH-FIRE-RESCUE INFORMATION

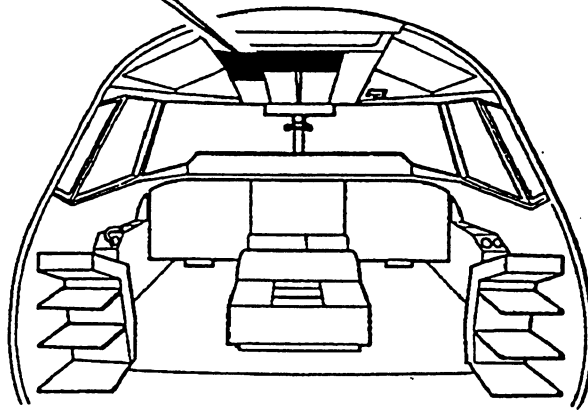


## ENGINE FIRE EXTINGUISHER OPERATION

- A. ELECTRICAL POWER MUST BE PRESENT
- B. PULL "PULL FUEL OFF" HANDLE.
- C. SELECT "EXTG" TOGGLE SWITCH TO EITHER "AFT BTL" OR "FWD BTL".

## REMOVING ELECTRICAL POWER

- A. SELECT "BATTERY MASTER" SWITCH TO "OFF".
- B. SELECT "AUX BATT" AND "MAIN BATT" SWITCHES TO "OFF".
- C. SELECT "EXT PWR" TO "OFF".



## ENGINE FIRE EXTINGUISHER AND BATTERY POWER SWITCH LOCATIONS